

NEA Scout Propulsion System

VACCO's cold gas Micro Propulsion System (MiPS) provides attitude control and orbital maneuvering. NASA's NEA Scout program utilizes VACCO's cold gas system to achieve highly reliable propulsion while observing an asteroid.

The VACCO NEA Scout MiPS is approximately 2U in volume and uses six 25 mN cold gas thrusters to develop 500 N-sec of total impulse that provides 37 m/s of delta-V for a 14 kg CubeSat. Each thruster independently operates to perform both delta-V and ACS maneuvers through an integrated microprocessor controller.

Performance density: 322 N-sec/L



SPACE

Features

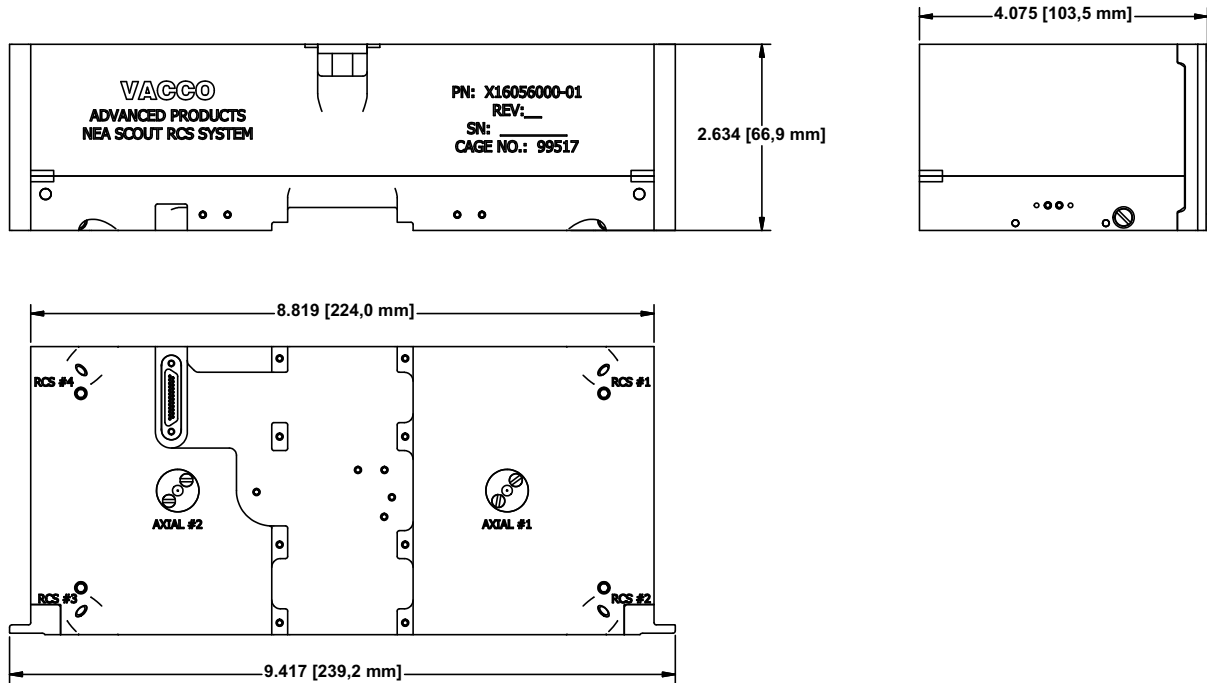
- Integral aluminum fluid control manifold and low friction, space grade valves
- All welded tank construction contains 1,280 g of propellant
- Integrated microcontroller and RS-422 interface enable high-level commands from the host spacecraft
- Low power with < 1 Watt for health and status monitoring
- Easily configured for different propellants
 - R-134a
 - R-236fa
- Performance density: 322 N-sec/L

Operating Parameters

Propellant.....R236fa	Total Impulse @10°C.....500 N-s
MDP.....6.89 Bar (100 psia)	Dry Mass.....1.263 kg Max
Proof Pressure.....10.34 Bar (150 psia)	Wet Mass 95% Fill @ 10°C.....2.540 kg Max
Burst Pressure.....13.79 Bar (200 psia)	Operating Voltage9.0-12.6 V _{DC}
Internal Leakage.....<0.5 scch R-236fa	Standby Power.....1.1 W Max
External Leakage.....<1.0 x 10 ⁻⁶ sccs GHe	Warmup Power.....55 W Max
Operating Temp.....-10°C to +55°C	Thruster Operating Power (4 thrusters).....9 W Max
Non-Operating Temp.....-24°C to +55°C	Data Interface.....RS-422

Performance characteristics are based on customer requirements. As such, they are not representative of component capabilities or limitations.

Envelope Drawing



Flow Schematic

